





AD

AD-E402 217

**Technical Report ARAED-TR-91013** 

# AN EXPLOSIVES PRODUCTS THERMODYNAMIC EQUATION OF STATE APPROPRIATE FOR MATERIAL ACCELERATION AND OVERDRIVEN DETONATION: THEORETICAL BACKGROUND AND FORMULATION

Ernest L. Baker

July 1991



# U.S. ARMY ARMAMENT RESEARCH, DEVELOPMENT AND ENGINEERING CENTER

Armament Engineering Directorate Picatinny Arsenai, New Jersey

Approved for public release; distribution is unlimited.

91-06254

**91** 7 28 048

The views, opinions, and/or findings contained in this report are those of the author(s) and should not be construed as an official Department of the Army position, policy, or decision, unless so designated by other documentation.

The citation in this report of the names of commercial firms of commercially available products or systems does not constitute official endorsement by or approval of the U.S. Government.

Destroy this report when no longer needed by any method that will prevent disclosure of contents or reconstruction of the document. Do not return to the originator.

## **REPORT DOCUMENTATION PAGE**

Form Approved OMB No. 0704-0188

needed, and completing and reviewing the collection of informatio	n. Send comments regarding this burden eat Operations and Reports, 1215 Jefferson Dai	imate or any other aspect	ructions, searching exisping data sources, gathering and maintaining the data of this collection of information, including suggestion; for reducing this burden, Artington, VA 22202-4302, and to the Office of Management and Budyet,
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE July 1991	3. RI	EPORT TYPE AND DATES COVERED
4. TITLE AND SUBTITLE AN EXPLOSIVES PRODUCTS EQ FOR MATERIAL ACCELERATION THEORETICAL BACKGROUND AN	UATION OF STATE APP AND OVERDRIVEN DE		5. FUNDING NUMBERS
6. AUTHOR(S) Ernest L. Baker			
7. PERFORMING ORSANIZATION NAME(S) AND ADDRESS(ES) ARDEC, AED ATTN: Energetic Materials and Warheads Div. ATTN: SMCAR-AEE-WW Picatinny Arsenal, NJ 07806-5000			PERFORMING ORGANIZATION     REPORT NUMBER  Technical Report ARAED-TR-91013
9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES) 10. SPONSORING/MONITORING AGENCY REPORT NUMBER STINFO Br ATTN: SMCAR-IMI-I Picatinny Arsenal, NJ 07806-5000			
11. SUPPLEMENTARY NOTES			
12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release, distribution is unlimited.			126. DISTRIBUTION CODE
detonations products expansion have been demonstrated to pro	<ul> <li>n. Common thermodyna ovide poor representation rmulated to provide a mo</li> </ul>	mic equations as of overdrive ore accurate re	ven detonation and lower pressure of state used for warheads design n detonation. The Jones-Wilkens- epresentation. This report provides a JWLB.
14. SUBJECT TERMS			15. NUMBER OF PAGES 17
			16. PRICE CODE  CLASSIFICATION 20. LIMITATION OF ABSTRACT
	THIS PAGE UNCLASSIFIED	OF ABSTR	ACT ASSIFIED SAR

### **CONTENTS**

	Page
Background	1
Equation of State Formulation	1
Speed of Sound	3
Adiabatic Gamma	4
Principal Isentrope Properties	4
Reacted Products Hugoniot	4
Conclusion	5
Symbols	5
References	6
Distribution List	9



Order toution/
A milactions Codes
Avail and/or Dist Special A-1

8 11 88261 Diff has

#### BACKGROUND

Many modern explosive applications require finit, element and finite difference modeling of both overdriven detonation and lower pressure detonation products expansion. Large detonation wave shaping and multiple point initiation are typical overdriven detonation Lower pressure detonation products expansion is required for material acceleration applications, such as explosively formed penetrators and shaped charges. Current thermodynamic equation of states used in dynamic finite element and finite difference programs are either parameterized to give agreement with thermochemical calculations [1] or experimental copper cylinder explosive expansion experiments [2,3]. calculations have proven to be very useful for the prediction of explosive products properties, particularly near and above the Chapman-Jouguet state [4,5]. Unfortunately, they do not reproduce the products expansion behavior accurately enough for typical warheads design. Currently, thermodynamic equations of state (JWL, Wilkens) [2,3] used for warheads design are normally calibrated to give agreement with copper cylinder explosive expansion experiments. These equations of state have not been calibrated for high pressures above the Chapman-Jouguet state. Experimentation [6] and comparison with thermochemical calculations (Figures 1,2, and 3) have demonstrated that a poor description of the high pressure region exists. In order to achieve a suitable equation of state for both overdriven and lower pressure products expansion, an appropriate equation of state form has been derived. A previous preliminary report [7] briefly describes the equation of state and parameterization methodology, but does not include details. This report provides a detailed theoretical background and equation of state formulation. A forthcoming report will provide a detailed description of the parameterization methodology.

#### EQUATION OF STATE FORMULATION

The equation of state form was chosen so as to adequately describe the high pressure regime produced by overdriven detonation, and yet retain the low pressure expansion behavior required for standard material acceleration modeling. To this end, the derived form is based on the Jones-Wilkens-Lee (JWL) equation of state [1] due to its computational robustness, and asymptotic approach to an ideal gas at high expansions. Additional exponential terms and a variable Gruneisen parameter have been added to adequately describe the high pressure region above the Chapman-Jouguet state. The resulting equation of state form, named Jones-Wilkens-Lee-Baker (JWLB), is

$$P = \sum_{i} A_{i} \left[ 1 - \frac{\lambda}{R_{i} V} \right] e^{-R_{i} V} + \frac{\lambda E}{V} + C \left( 1 - \frac{\lambda}{\omega} \right) V^{(\omega + 1)}$$
(1)

where,

$$\lambda = \sum_{i} (A_{\lambda i} V + B_{\lambda i}) e^{-R} \lambda i^{V} + \omega$$
 (2)

For consistency with the JWL equation of state, V is defined as a specific volume ratio, V =

 $\rho_0/\rho$  and E is defined as  $E = \rho_0 e$  where e is the specific internal energy. The JWLB equation of state form is based on a first order expansion around the principal isentrope:

$$P_{\bullet} = \sum_{i} \hat{A}_{i} e^{-R} i^{V} + CV^{(\omega+1)}. \tag{3}$$

Using the Gruneisen Parameter,

$$\lambda = V \frac{\partial P}{\partial E} \bigg|_{V}, \tag{4}$$

the isentropic identity,

$$P = -\frac{\partial E}{\partial V}\Big|_{S} \tag{5}$$

$$\Rightarrow E_{\bullet} - E_{cj} = \sum_{i=1}^{A_i} e^{-R_i V} + \frac{C}{\omega} V^{-\omega} - \sum_{i=1}^{A_i} e^{-R_i V_{cj}} - \frac{C}{\omega} V_{cj}^{-\omega}, \tag{6}$$

and the Chapman-Jouguet condition,

$$E_{cj} = E_o + \frac{1}{2}(P_{cj} + P_o)(V_o - V_{cj})$$

$$= \sum_{i} \frac{A_i}{R_i} e^{-R_i V_{cj}} + \frac{C}{\omega} V_{cj}^{-\omega}, \qquad (7)$$

the final form may be derived

$$P = \frac{\lambda}{V}(E - E_s) + P_s$$

$$= \frac{\lambda}{V} \left(E - \sum_{i=1}^{A_i} e^{-R_i V} - \frac{C}{\omega} V^{-\omega}\right) + \sum_{i=1}^{A_i} e^{-R_i V} + CV^{(\omega+1)}$$
(8)

⇒ Final Form.

Some important characteristics of the equation of state are that the Gruneisen Parameter  $\lambda$ , is represented as an analytic function of specific volume, V.  $\lambda + 1$  approaches a constant adiabatic gamma,  $\frac{V}{P} \frac{\partial P}{\partial V}\Big|_{S} = \omega + 1$  for large V, so that ideal gas behavior is asymptotically

approached. The principal isentrope description is essentially identical to JWL, with the exception of an increased number of exponential terms. It has been found that for most explosives, three exponential terms (instead of two in JWL) are adequate to describe the principal isentrope over both the high pressure region above the Chapman-Jouguet state and the lower pressure expansion region. It is important to note that the internal energy referencing is defined by (7). The value of  $E_0$  must be consistent so that,

$$E_{\rm o} = \sum_{i=1}^{A_{i}} e^{-R_{i} V_{\rm cj}} + \frac{C}{\omega} V_{\rm cj}^{-\omega} - \frac{1}{2} (P_{\rm cj} + P_{\rm o}) (V_{\rm o} - V_{\rm cj}) . \tag{9}$$

Normally, the equation of state parameters are chosen so that  $E_{\circ}$  has the value  $E_{\circ} = \rho_{\circ} \Delta H$ , where  $\Delta H$  is the heat of detonation. This is consistent with the initial internal energy of the unreacted material having a value of zero.

#### SPEED OF SOUND

The generalized speed of sound is often required for implementation into a finite element or finite difference program. The speed of sound may be easily derived as follows,

$$\frac{\partial P}{\partial V} = \sum_{i} A_{i} \left[ \frac{\lambda}{R_{i} V^{2}} - \frac{\frac{\partial \lambda}{\partial V}}{R_{i} V} \right] e^{-R_{i} V} - \sum_{i} A_{i} \left[ 1 - \frac{\lambda}{R_{i} V} \right] R_{i} e^{-R_{i} V} 
+ C \left( -\frac{\frac{\partial \lambda}{\partial V}}{\omega} \right) V^{(\omega+1)} - C(\omega+1) (1 - \frac{\lambda}{\omega}) V^{(\omega+2)} 
- \frac{E\lambda}{V^{2}} + \frac{\lambda}{V} \frac{\partial E}{\partial V} + \frac{E}{V} \frac{\partial \lambda}{\partial V} \tag{10}$$

$$(5), (10) \Rightarrow \frac{\partial P}{\partial V} \Big|_{S} = \sum_{i} A_{i} \left[ \frac{\lambda}{R_{i} V^{2}} - \frac{\frac{\partial \lambda}{\partial V}}{R_{i} V} - R_{i} + \frac{\lambda}{V} \right] e^{-R_{i} V}$$

$$- C \left[ (\omega + 1)(1 - \frac{\lambda}{\omega}) + V \frac{\frac{\partial \lambda}{\partial V}}{\omega} \right] V^{-(\omega + 2)}$$

$$- \frac{E\lambda}{V^{2}} - \frac{\lambda}{V} P + \frac{E}{V} \frac{\partial \lambda}{\partial V}$$

$$(11)$$

$$c^{2} = \frac{\partial P}{\partial \rho} \Rightarrow \rho_{0}c^{2} = -V^{2}\frac{\partial P}{\partial V}\Big|_{S}$$
(12)

$$\Rightarrow \rho \omega^{2} = \sum_{i} A_{i} \left[ V \frac{\partial \lambda}{\partial V} - \frac{\lambda}{R_{i}} + R_{i} V^{2} - \lambda V \right] e^{-R_{i} V}$$

$$+ C \left[ (\omega + 1)(1 - \frac{\lambda}{\omega}) + V \frac{\partial \lambda}{\partial V} \right] V^{-\omega} + E\lambda + \lambda VP - EV \frac{\partial \lambda}{\partial V}$$
(13)

$$\frac{\partial \lambda}{\partial V} = \sum_{i} A_{\lambda i} e^{-R_{\lambda i} V} - \sum_{i} (A_{\lambda i} V + B_{\lambda i}) R_{\lambda i} e^{-R_{\lambda i} V}$$
(14)

#### ADIABATIC GAMMA

Another useful quantity is the generalized adiabatic gamma. The adiabatic gamma may also be easily derived,

$$\gamma = -\frac{\partial \ln P}{\partial \ln V} \bigg|_{S} = -\frac{V}{P} \frac{\partial P}{\partial V} \bigg|_{S} \tag{15}$$

$$(11), (15) \Rightarrow \gamma = \frac{1}{P} \sum_{i} A_{i} \left[ \frac{\frac{\partial \lambda}{\partial V}}{R_{i}} - \frac{\lambda}{R_{i}V} \frac{\frac{\partial \lambda}{\partial V}}{R_{i}} + R_{i}V - \lambda \right] e^{-R_{i}V}$$

$$+ \frac{C}{P} \left[ (\omega + 1)(1 - \frac{\lambda}{\omega}) + V \frac{\frac{\partial \lambda}{\partial V}}{\omega} \right] V^{(\omega + 1)} + \frac{E\lambda}{VP} + \lambda - \frac{E}{P} \frac{\partial \lambda}{\partial V}$$

$$(14) \Rightarrow \frac{\partial \lambda}{\partial V} = \sum_{i} A_{\lambda i} e^{-R} \lambda_{i}^{V} - \sum_{i} (A_{\lambda i}V + B_{\lambda i}) R_{\lambda i}^{2} e^{-R} \lambda_{i}^{V}.$$

$$(16)$$

#### PRINCIPAL ISENTROPE PROPERTIES

Often, properties along the isentrope that passes through the Chapman-Jouguet state are of particular interest. The Gruneisen parameter is given by (2). The adiabatic gamma is given by,

$$(3) \Rightarrow \frac{\partial P_s}{\partial V} = -\sum_i A_i R_i e^{-R_i V} - C(\omega + 1) V^{(\omega + 2)}$$
(17)

(3), (15) 
$$\Rightarrow \gamma_s = \frac{V \sum_{i} A_i R_i e^{-R_i V} + C(\omega + 1) V^{(\omega + 1)}}{\sum_{i} A_i e^{-R_i V} + CV^{(\omega + 1)}}.$$
 (18)

The speed of sound along the principal isentrope is given by,

$$(12) \Rightarrow \rho_0 c^2 = \sum_i \Lambda_i R_i V^2 e^{-R_i V} + C(\omega + 1) V^{(\omega)}.$$

$$(19)$$

#### REACTED PRODUCTS HUGONIOT

Another often used state space locus is the reacted products Hugoniot. Assuming the initial pressure to be zero, conservation gives,

Mass: 
$$\rho D = \rho(D - u)$$
 (20)

Momentum: 
$$P = \rho_0 Du$$
 (21)

Energy: 
$$\frac{D^2}{2} + \frac{E_0}{\rho_0} = \frac{P}{\rho} + \frac{(D - u)^2}{2} + \frac{E}{\rho_0}$$
 (22)

$$(20) \Rightarrow u = \frac{\rho - \rho_c}{\rho} D = (1 - V)D \tag{23}$$

(23),(20) 
$$\Rightarrow P = \rho_0(1-V)D^2 \Rightarrow D^2 = \frac{P}{\rho_0(1-V)}$$
 (24)

(22), (24) 
$$\Rightarrow \frac{P}{2\rho_{o}(1-V)} + \frac{E_{o}}{\rho_{o}} = \frac{P}{\rho} + V^{2} \frac{P}{2\rho_{o}(1-V)} + \frac{E}{\rho_{o}}$$
 (25)

$$\Rightarrow E = E_{\circ} + \frac{P}{2(1-V)} - V^2 \frac{P}{2(1-V)} - PV$$

$$= E_o + \frac{P}{2(1-V)}(1-V^2-2V+2V^2) = E_o + \frac{P}{2(1-V)}(V^2-2V+1)$$

$$= E_o + \frac{P(1-V)}{2}$$
(26)

$$(1),\,(26)\Rightarrow E\left(1-\frac{\lambda(1-V)}{2V}\right)$$

$$= E_{\rm e} + \left( \sum_{i} A_{i} \left[ 1 - \frac{\lambda}{R_{i} V} \right] e^{-R_{i} V} + C \left( 1 - \frac{\lambda}{\omega} \right) V^{(\omega + 1)} \right) \frac{(1 - V)}{2}$$
(27)

$$\Rightarrow E = \frac{E_0 + \left(\sum_{i} A_i \left[1 - \frac{\lambda}{R_i V}\right] e^{-R_i V} + C(1 - \frac{\lambda}{\omega}) V^{(\omega+1)}\right) \frac{(1-V)}{2}}{\left(1 - \frac{\lambda(1-V)}{2V}\right)}$$
(28)

#### CONCLUSION

An advanced transmodynamic equation of state applicable to problems involving overdriven detonation and material acceleration has been developed and calibrated for several explosives. This manuscript has presented a theoretical background and formulation for the new equation of state. A formcoming report will describe the parameterization methodology. The new equation of state maintains required low pressure expansion behavior while providing a better high pressure description applicable to overdriven detonation. Typical applications of the new equation of state include wave shaped and peripherally initiated munitions. Finally, the equation of state asymptotically approaches the constant gamma equation of state at high expansions. It is therefore believed that the equation of state will be applicable to extremely large volume expansion applications, such as blast, without a loss of accuracy in the higher pressure regions. In order to use the equation of state for very large expansions, proper calibration to very large volume expansion experimentation will be required.

#### **SYMBOLS**

P = Pressure

 $\rho = Density$ 

V = Ratio of Specific Volume to Initial Specific

#### Volume

- E = Specific Internal Energy divided by Initial Specific Volume
- $\lambda = Gruneisen Parameter$
- γ = Adiabatic Gamma
- c = Sound Speed
- D = Detonation Velocity
- u = Mass Velocity
- $\Delta H = \text{Heat of Detonation}$

 $A_i$ ,  $R_i$ , C,  $A_{\lambda i}$ ,

 $B_{\lambda i}$ ,  $R_{\lambda i}$ ,  $\omega = Equation of State Constants$ 

- . ≡ Isentropic
- cj ≡ Chapman-Jouguet State
- u = Initial Conditions

#### REFERENCES

- 1. C.L. Mader, Numerical Modeling of Detonations, University of California Press, 1979.
- 2. M.L. Wilkens, "The Equation of State of PBX 9404 and LX04-01", Lawerence Radiation Laboratory, Livermore, Rept. UCRL-7797, 1964.
- 3. E.L. Lee, H.C. Horning, and J.W. Kury, "Adiabatic Expansion of High Explosive Detonation Products", University of California Report No. UCRL-50422, 1968.
- 4. C.L. Mader, "FORTRAN BKW A Code for Computing the Detonation Properties of Explosives", University of California Technical Report LA-3704, 1967.
- 5. M. Cowperthwaite and W.H. Zwisler, "TIGER Computer Program Documentation", Stanford Research Institute Publication No. Z106.
- 6. L.G. Green, C.M. Tarver, and D.J. Erskine, "Reaction Zone Structure in Supracompressed Detonating Explosives", University of California Report No. UCRL-101862.
- 7. E.L. Baker, and J. Orosz, "Advanced Warhead Concepts: An Advanced Thermodynamic Equation of State for Overdriven Detonation", Picatinny Arsenal Technical Report ARAED-TR-91007, 1991.

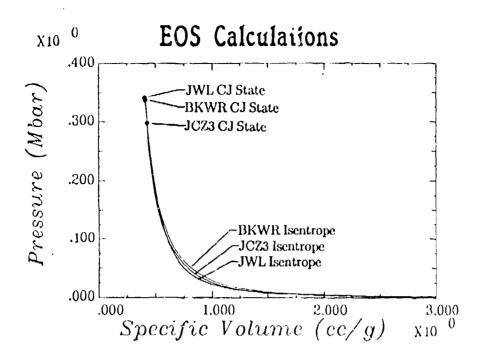


Figure 1. Pressure versus specific volume for the principal isentrope of octol 75/25 below the Chapman-Jouguet state. The thermochemical calculations (BKWR and JCZ3) agree fairly well with the standard JWL.

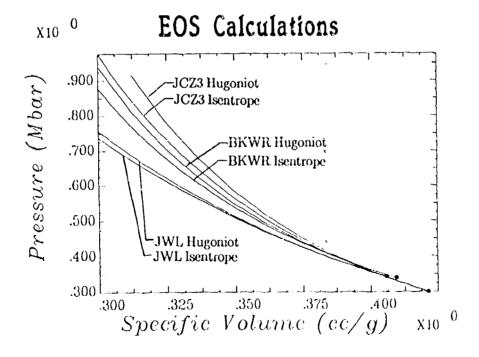


Figure 2. Pressure versus specific volume for the principal isentrope and reactive Hugoniot of octol 75/25 above the Chapman-Jouguet state. The standard JWL underpredicts the high pressure region.

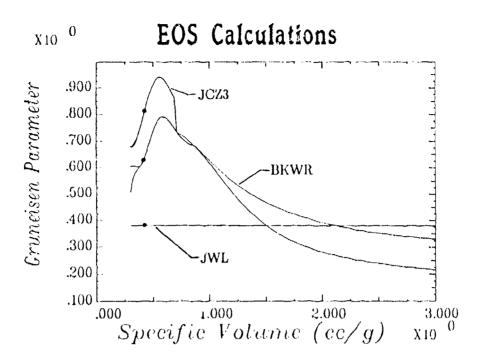


Figure 3. The Gruneisen parameter versus specific volume for the principal isentrope and reactive Hugoniot.

#### **DISTRIBUTION LIST**

Commander

U. S. Army Armament Research, Development and Engineering Center

ATTN: SMCAR-IMI-1 (5)

SMCAR-CO

SMCAR-TD

SMCAR-TDC

SMCAR-AE (3)

SMCAR-AEE (3)

Picatinny Arsenal, NJ 07806-5000

Commander

U. S. Army Armament Research, Development and Engineering Center

ATTN: AMSMC-GCL(D)

AMSMC-DSM-B(D)

Picatinny Arsenal, NJ 07806-5000

Commander

U. S. Army Materiel Command

ATTN: AMCCN-C

AMCLD,D. Vitali

5001 Eisenhower Avenue

Alexandria, VA 22333-0001

Commander

U. S. Army Armament, Munitions and Chemical Command

ATTN: AMSMC-DCG

AMSMC-DSM-A

Aberdeen Proving Ground, MD 21010-5423

Commander

U. S. Army Depot Systems Command

ATTN: AMSDS-QV

AMSDS-T

AMSDS-SM-SPA

AMSDS-CG

Chambersburg, PA 17201-4170

Administrator

Defense Technical Information Center

ATTN: Accessions Division (2)

Cameron Station

Alexandria, VA 22304-6145

Director

U. S. Army Materiel Systems Analysis Activity

ATTN: AMXSY-MP

AMXSY-RW

Aberdeen Proving Ground, MD 21005-5066

Commander

Chemical Research, Development and Engineering Center

U. S. Army Armament, Munitions and Chemical Command

ATTN: SMCCR-SPS-IL

SMCCR-CO SMCCR-MU SMCCR-SPS-M SMCAR-RSP-A

Aberdeen Proving Ground, MD 21010-5423

Director

Ballistic Research Laboratory

ATTN: AMXBR-OD-ST

SLCBR-TB-EE

Aberdeen Proving Ground, MD 21005-5066

Chief

Benet Weapons Laboratory, CCAC

Armament Research, Development and Engineering Center U. S. Army Armament, Munitions and Chemical Command ATTN: SMCAR-CCB-TL

Watervliet, NY 12189-5000

Commander

U. S. Army Armament, Munitions and Chemical Command

ATTN: SMCAR-FSP-D

**SMCAR-DS** 

AMSMC-IMF-L

Rock Island, IL 61299-6000

Director

U. S. Army TRADOC Systems Analysis Activity

ATTN: ATAA-SL

White Sands Missile Range, NM 88002

Commander

Naval Weapons Center

ATTN: L. Smith, Code 320

A. Amster, Code 385

R. Reed, Jr., Code 388

K. J. Graham, Code 3835

China Lake, CA 93555

Director

Lawrence Livermore National Laboratory

University of California

P.O. Box 808

ATTN: M. Finger

C. Tarver

L. Green

Livermore, CA 94550

Director

Los Alamos National Laboratory

ATTN: L. Hull

T. Bennion

J. Chapyak

J. Ritchie

Los Alamos, NM 87545

Director

Sandia National Laboratory Albuquerque, NM 87185

Commander

Naval Surface Weapons Center

ATTN: Code G13

Dahlgreen, VA 22448

Commander

Naval Surface Weapons Center

ATTN: L. Roslund, R122

M. Stosz, R 121

Code X211, Library

E. Zimet, R13

R.R. Bernecker, R13

J.W. Forbes, R13

S.J. Jacobs, R10

C.Dickinson

J. Short, R12

Silver Spring, MD 20910

Air Force Armament Laboratory

ATTN: J. Forster, AFATL-MNW

G. Parson, AFATL-MNE

Eglin AFB,FL 32542-5434

Dyna East Corporation

ATTN: William Flis

3201 Arch Street

Philadelphia, PA 19104

Aerojet Ordnance Company

ATTN: Dan Donati

2521 Michelle Dr

Tustin, CA 92680

Applied Ordnance Technology, Inc.

ATTN: NIMIC, Eddie Norton

1000 Century Plaza, Suite 212, BX24

10630 Little Patuxent Parkway

Columbia, MD 21044

#### Commander

Chemical Research, Development and Engineering Center U.S. Army Armament, Munitions and Chemical Command ATTN: SMCCR-MSI Aberdeen Proving Ground, MD 21010-5423

Royal Ordnance

ATTN: Dr. Peter R. Lee

Westcott, Aylesbury

Buckinghamshire, HP180NZ England